

SUPPLEMENTAL TYPE CERTIFICATE

10041707 REV. 2

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

BERINGER AERO S.A.S.

**AERPOLE
05130 TALLARD
FRANCE**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: 1. EASA.A.049
2. EASA.A.274

Type Certificate Holder: SCHEMPP HIRTH FLUGZEUGBAU GMBH

Type: DISCUS A, VENTUS A

Model: 1. DISCUS A, DISCUS 2A
2. VENTUS A, VENTUS 2A

Description of Design Change:

Installation of main wheels and brake system P/N : JA-02.

Revision 1 : Update with new wheel P/N : RF-007(B) with brake EA-006E supersedes the P/N : JA-02.

Revision 2 : Update with new wheel P/N : RF-020 supersedes the wheel P/N : RF-007.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 28 September 2017



Dominique ROLAND
Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)

Associated Technical Documentation:

Flight manual supplement FM-STC-007(0) dated 12/09/2012
or later revisions of the above listed documents approved by EASA.

Modification Dossier: DM-STC-007(2) dated 30/07/2017.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -