

## SUPPLEMENTAL TYPE CERTIFICATE

**10029629 REV. 5**

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### **BERINGER AERO S.A.S.**

**AEROPOLE  
05130 TALLARD  
FRANCE**

EASA.AP215

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

**Type Certificate Number:** CH F 56-10

**Type Certificate Holder:** PILATUS AIRCRAFT LTD

**Type:** PC6

**Model:** PC-6/A2-H2, PC-6/B1-H2

PC-6/B2-H2, PC-6/B2-H4

**Description of Design Change:**

Installation of Beringer brakes.

Revision 1: Update with an optional small main wheel.

Revision 2: Update with an optional axle, update of manuals.

Revision 3: Update with an additional brake disc.

Revision 4: Update with an additional brake pad.

Revision 5: Update with wear limit modification.

**EASA Certification Basis:**

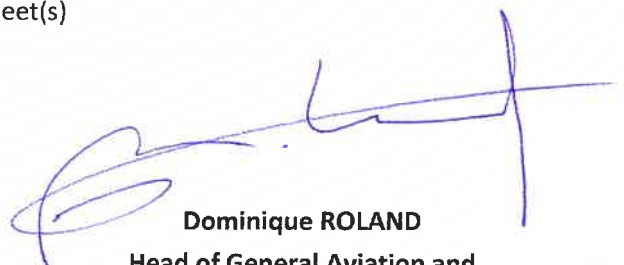
The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

**For the European Aviation Safety Agency**

**Cologne, Germany, 06 June 2018**



**Dominique ROLAND**  
**Head of General Aviation and**  
**Remotely Piloted Aircraft Systems (RPAS)**

**Associated Technical Documentation:**

Modification file DM-STC-02 Rev 5 dated 12/05/2018.

**Limitations/Conditions:**

The approval holder shall fulfill the obligations of Part 21, Point 21.A.109 or 21.A.451(b) (as applicable).

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -