

SUPPLEMENTAL TYPE CERTIFICATE

10043346 REV. 5

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

BERINGER AERO S.A.S.

**AEROPOLE
05130 TALLARD
FRANCE**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.A.007

Type Certificate Holder: CIRRUS DESIGN CORPORATION

Type: CIRRUS SR20, SR22

Model: SR20, SR22, SR22T

Description of Design Change:

Installation of main wheels (RF-006(A)), nose wheel (RA-002(A)) and brake system (EA-003.4N(A))

Revision 1: Upgrade of the brake system: EA-003.4N(B) supersedes EA-003.4N(A)

Revision 2: Update of maintenance manual and part list

Revision 3: Maintenance manuals updates and introduction of an aircraft flight manual supplement (AFMS)

Revision 4: Update with new wheel part Number: RA-014 which supersedes RA-002(B)

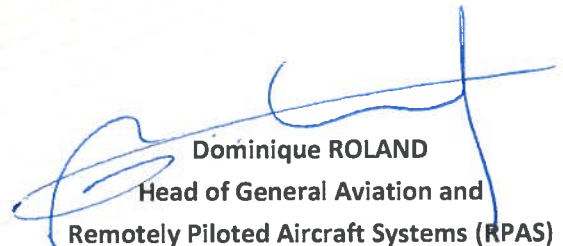
EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 05 September 2017



Dominique ROLAND
Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)

10017320

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The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

-AFM Supplement ref AFMS-STC-004(0) dated 20/10/2015

-Maintenance and Overhaul Manual including Instructions for Continued Airworthiness ref MM-STC-004(4) dated 05/06/2017

or later revisions of the above listed documents approved by EASA.

Modification Dossier ref DM-STC-004(5) dated 31/05/2017

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -