

SUPPLEMENTAL TYPE CERTIFICATE

10065112 REV. 3

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

BERINGER AERO

30 RUE PIERRE GEORGES LATECOERE
05130 TALLARD
FRANCE

EASA.AP215

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.IM.A.077

Type Certificate Holder: PIPER AIRCRAFT, INC.

Type: PA-46

Model: PA-46-310P, PA-46-500TP

PA-46R-350T, PA-46-350P

Description of Design Change:

Servicing instruction & SensAIR option

Revision 2 of this STC introduces an adapter for the tow bar

Revision 3 is an administrative change and addition of SensAIR option

EASA Certification Basis:

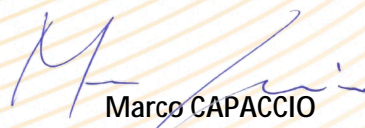
The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 16 March 2022


Marco CAPACCIO
Section Manager

Small Aircraft, Balloons & Airships



Associated Technical Documentation:

- Installation and Instructions for Continued Airworthiness: ICA-STC-017 rev. 4 dated 20 December 2021;
- Project specific Certification Plan: DM-MOD002-STC-017 rev 00 dated 17 February 2022;

or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -