

SUPPLEMENTAL TYPE CERTIFICATE

10057960 REV. 5

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

BERINGER AERO

30 RUE PIERRE GEORGES LATECOERE 05130 TALLARD FRANCE

EASA.AP215

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.IM.A.234

US 2A13

Type Certificate Holder: PIPER AIRCRAFT, INC.

Type: PA-28

Model: PA-28-140 (CHEROKEE CRUISER)

PA-28-150 (Cherokee)

PA-28-151 (Cherokee Warrior)

PA-28-160 (Cherokee)

PA-28-161 (Warrior II)

PA-28-161 (Warrior III)

PA-28-180 (Archer)

PA-28-180 (Cherokee)

PA-28-181 (Archer II)

PA-28-181 (Archer III)

Description of Design Change:

Replacement of Wheel and Brake System

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 18 March 2022

Marco CAPACCIO Section Manager

Small Aircraft, Balloons & Airships



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Revision 4: Addition of PA-28-140 model

Revision 5: Update with new servicing instructions within ICA document and addition of SensAIR device as an option

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

Modification file: DM-STC-014 rev 07 dated 11 Mar. 2022; Installation and Instructions for Continued Airworthiness: ICA-STC-014 rev 07 dated 19 Oct. 2021; or later revisions of the above listed documents approved by EASA.

Limitations/Conditions:

The approval holder shall fulfill the obligations of Part 21, Paragraph 21A.109.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.



