

# SUPPLEMENTAL TYPE CERTIFICATE

## 10052988 REV. 2

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### BERINGER AERO

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EASA.AP215

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

**Type Certificate Number:** EASA.A.083

**Type Certificate Holder:** JSC "Sportinè Aviacija ir KO"

**Type:** LAK-17

**Model:** LAK-17A

LAK-17AT

LAK-17B FES

#### Description of Design Change:

Installation of main wheels and brake system P/N : RF-002(B) and EA-006

Revision 1: Update of part numbers including the new wheel P/N : RF-019(B) which supersedes the P/N : RF-002(B)

Revision 2: Update with new servicing instructions within ICA document and addition of SensAIR device as an option

#### EASA Certification Basis:

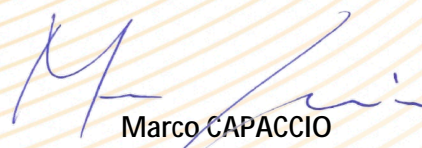
The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

**For the European Union Aviation Safety Agency**

Cologne, Germany, 18 March 2022



Marco CAPACCIO

Section Manager

Small Aircraft, Balloons & Airships





**Associated Technical Documentation:**

Modification file: DM-STC-011 rev 02 dated 11 Mar. 2022;  
Installation and Instructions for Continued Airworthiness: ICA-STC-011 rev 02 dated 19 Oct. 2021;  
or later revisions of the above listed documents approved by EASA.

**Limitations/Conditions:**

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

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