

SUPPLEMENTAL TYPE CERTIFICATE

10052286 REV. 5

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

BERINGER AERO

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EASA.AP215

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.A.362

Type Certificate Holder: EXTRA Flugzeugproduktions-

Type: EA 300

Model: EA 300, EA 300/200

EA 300/L, EA 300/LC

EA 300/LT, EA 300/S

EA 300/SC

Description of Design Change:

Replacement of wheel and brake system

Rev.1 - Update with longer axle bolts

Rev.2 - Update with different hydraulic fitting

Rev.3 - Test Report Improvement

Rev.4 - Update with new wheel part number

Rev.5 - Update with new servicing instructions within ICA document and addition of SensAIR device as an option

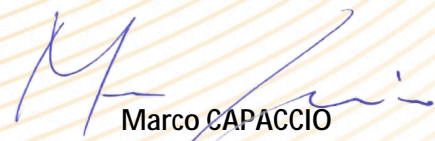
EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 18 March 2022



Marco CAPACCIO
Section Manager

Small Aircraft, Balloons & Airships



The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

Modification file: DM-STC-012 rev 06 dated 11 Mar. 2022;
Installation and Instructions for Continued Airworthiness: ICA-STC-012 rev 04 dated 19 Oct. 2021;
or later revisions of the above listed documents approved by EASA.

Limitations/Conditions:

The approval holder shall fulfill the obligations of Part 21, Point 21.A.109 or 21.A.451(b) (as applicable).

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -